



Research paper

Effects of asymmetrical damping on a 2 DOF quarter-car model under harmonic excitation

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ABSTRACT

The objective of this work is to study the dynamical behavior of vehicle suspension systems employing asymmetrical viscous damping, with a focus on improving passenger comfort. Previous studies have shown that the use of asymmetrical dampers in these types of systems can be advantageous with regard to comfort of the passengers. The modeling and the behavior of a quarter-car model with asymmetrical viscous damping under harmonic excitation is presented. The response is obtained with an analytical approximation via the method of Harmonic Balance. The choice of the asymmetry ratio diminishes the effects that the uneven road causes on the displacement and acceleration of the sprung mass. Although current systems usually adopt larger damping during the expansion phase, it is shown in this work that, for lower frequencies, smaller damping in this phase results in better comfort.

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1. Introduction

Suspension systems with nonlinear elements have been studied extensively, including nonlinear springs and nonlinear dampers [1–4]. Shekhar *et al.* showed that the dampers have more influence on the dynamics than the springs [5], and the difference between linear and nonlinear elements is more pronounced for more severe shock loads [6]. These nonlinear elements have been included in active and semi-active suspension systems, in which damping properties vary according to a control algorithm applied, adding or removing energy of the system by means of valves with variable orifice area or magneto-rheological fluids [7–10]. Electromechanical suspension systems have also been studied with great interest as a means of energy recovering system [11,12].

Asymmetrical dampers are usually designed to provide significantly greater damping force during the extension period compared to the force during the compression period [13]. These different damping coefficients in extension and compression define an asymmetry ratio. The use of an asymmetrical damping induces a shift in the steady state response of the system. The magnitude of this shift depends on the asymmetry ratio and on the excitation frequency [14].

In addition to vertical motion, usually studied with quarter-car models, the pitch displacement and acceleration were shown to be smoother for a half-car model which utilizes asymmetrical dampers [15]. Also, the vertical and angular accelerations reach minimum values at different points in the presence of asymmetrical dampers. Generally, the extension force is

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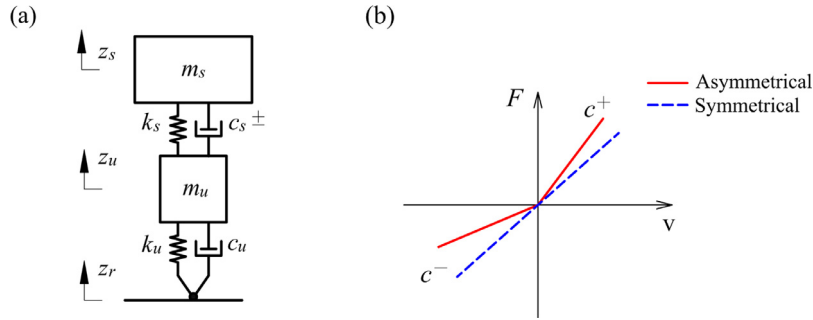


Fig. 1. Two-degree-of-freedom quarter-car model of a suspension system (a) and the generic asymmetrical characteristic of a shock absorber (b).

defined between three to four times the compressive force [16]. However, the variation of this ratio can provide very different behaviours, and has been used in control strategies [17]. The use of more stages of asymmetry has also been studied [18].

It is commonly understood and accepted that the human response to dynamic excitation depends on many mechanical, physical, physiological and psychological parameters [19]. Besides the suspension system *per se*, the seat of the vehicle is also focus of study to improve comfort in vehicles [20]. The level of comfort for the passengers depends on factors as the frequency of vibration, direction, location and time of exposure to which the passenger is subjected [20]. Apart from causing discomfort, severe vibrations may cause occupational disorders [21].

In order to further explore the use of asymmetrical damping to improve the level of comfort at different frequencies of excitation, in this paper a classical quarter-car model with 2 degrees of freedom is used with harmonic base excitation. The model is presented in Section 2; the response of the system is obtained using an analytical approximation via the method of Harmonic Balance, shown in Section 3; the convergence of the approximate solutions is illustrated in Section 4; the influence of the asymmetry ratio on the response is studied in Section 5; and conclusions are drawn in Section 6.

2. A 2-DOF quarter-car model with asymmetrical damper

There are many models suited for the dynamical analysis of vehicles, varying from simpler models to more sophisticated ones. Lower order models with lumped parameters are very common for analysis of the vertical dynamics. Quarter-car models describe exclusively vertical motion of the chassis.

Fig. 1 shows a two-degree-of-freedom shock absorption system. This system comprehends two subsystems connected in series. Each subsystem comprehends a spring and a viscous damper assembled on a structure with mass.

The base of the system represents the road on which the vehicle travels, whose irregularities cause the forces that excite the system. The lower subsystem, comprehended by the mass m_u , the spring with elastic constant k_u and the viscous damper with viscous damping coefficient c_u , represents the un-sprung mass of the vehicle, e.g. wheels, tyres, brake discs, uprights. These elements have elastic and damper characteristics that, although not being the proper suspension system, contribute to the overall dynamical behaviour of the vehicle. The upper subsystem, comprehended by m_s , k_s , and c_s , represents the proper suspension system and the sprung mass. The nonlinear absorber is adopted at this subsystem. It is clear that the behaviour of the whole system depends on the contribution of each subsystem, as on the interaction between them.

According to the classical theory of vibrations [22,23], the equation of motion for the system depicted in Fig. 1, regarding vertical displacement relative to the trivial equilibrium positions, can be written as:

$$\begin{aligned} m_u \ddot{z}_u + f_{cu} + f_{k_u} - f_{c_s} - f_{k_s} &= 0 \\ m_s \ddot{z}_s + f_{c_s} + f_{k_s} &= 0 \end{aligned} \tag{1}$$

The springs are considered linear throughout this study. The spring forces f_{k_u} and f_{k_s} are proportional to the relative displacement, and are given by:

$$f_{k_u} = k_u(z_u - z_r) \qquad f_{k_s} = k_s(z_s - z_r) \tag{2}$$

The linear viscous damping force f_{cu} is proportional to the relative velocity, and is given by:

$$f_{cu} = c_u(\dot{z}_u - \dot{z}_r) \tag{3}$$

The nonlinear damping force depends on the relative velocity and its direction. In order to perform the switching of the absorber signal between the viscous damping coefficient at expansion (c_s^+) and at compression (c_s^-), representing its nonlinear characteristic, a Heaviside type function is used as follows:

$$c_s^\pm (\dot{z}_s - \dot{z}_u) = \begin{cases} c_s^+ & \text{if } \dot{z}_s - \dot{z}_u \geq 0 \\ c_s^- & \text{if } \dot{z}_s - \dot{z}_u < 0 \end{cases} \tag{4}$$

The base excitation is a result from the irregularities of the road on which the vehicle travels. The base excitation used here is a harmonic function given by

$$z_r(t) = z_{\max} \sin(\omega_r t + \phi) \tag{5}$$

in which z_{\max} is the amplitude of excitation, ω is the frequency of excitation and ϕ is the phase. The amplitude and frequency of excitation can be tuned to represent different types of roads and different velocities at which the vehicle is travelling. Usually ϕ can be set arbitrarily to zero. However, in this case it has to be taken into consideration for the correct match between the responses of the expansion and compression phases.

It is convenient to rewrite Eq. (1) in terms of the relative displacements between unsuspended mass and base displacements, and between suspended mass and unsuspended mass displacements, respectively given by

$$y_u = z_u - z_r \qquad y_s = z_s - z_u \tag{6}$$

Using equations (6), the equations of motion in terms of relative displacements are given by

$$\begin{aligned} m_u \ddot{y}_u + m_u \ddot{z}_r + c_u \dot{y}_u + k_u y_u - c_s^\pm \dot{y}_s - k_s y_s &= 0 \\ m_s \ddot{y}_s - \frac{m_s}{m_u} c_u \dot{y}_u - \frac{m_s}{m_u} k_u y_u + \left(1 + \frac{m_s}{m_u}\right) c_s^\pm \dot{y}_s + \left(1 + \frac{m_s}{m_u}\right) k_s y_s &= 0 \end{aligned} \tag{7}$$

The nondimensional time, mass ratio, stiffness ratio and damping factors for the unsuspended and suspended systems are introduced, respectively, as

$$\tau = \sqrt{\frac{k_s}{m_s}} t \qquad m = \frac{m_s}{m_u} \qquad k = \frac{k_u}{k_s} \qquad \zeta_u = \frac{c_u}{2\sqrt{k_s m_s}} \qquad \zeta_s^\pm = \frac{c_s^\pm}{2\sqrt{k_s m_s}} \tag{8}$$

Using Eqs. (7) and (8), the nondimensional equations of motion in terms of relative displacement are given by

$$\begin{aligned} \frac{1}{m} y_u'' + \frac{1}{m} z_r'' + 2\zeta_u y_u' + k y_u - 2\zeta_s^\pm y_s' - y_s &= 0 \\ y_s'' - 2m\zeta_u y_u' - m k y_u + (1 + m) 2\zeta_s^\pm y_s' + (1 + m) y_s &= 0 \end{aligned} \tag{9}$$

in which a prime denotes differentiation with respect to τ .

3. Approximate analytical solution using Harmonic Balance

The system represented by (9) possesses a non-smooth nonlinearity due to the asymmetric damper and can be classified as a piecewise linear system. A semi-analytical exact solution can be obtained by patching together the complete solutions in the two different linear systems with numerical solution of the switching times. However, the algebra becomes involved due to the complicated analytical expressions for the complete coupled system. Since numerical solution of the switching times between the two linear systems is anyway involved in the exact formulation, a closed form semi-analytical approximate solution based on harmonic balance is sought.

The periodic solution for y_s and y_u for a given harmonic forcing is approximated as a Fourier series, with the fundamental frequency equal to the excitation frequency. The Fourier coefficients for the approximate solution can be obtained in closed form in terms of the switching times, and numerical solution of the switching times from an algebraic equation is required for approximation involving higher harmonics, although the one-term approximation leads to a closed-form analytical solution. The accuracy of this approximate solution improves with increasing number of harmonics retained in the approximation, as will be shown later. For most practical scenarios, an approximate solution with the first three harmonics is as good as the exact solution. First, the closed form analytical expression obtained for the one-term approximation is presented, followed by a refined solution with more harmonics.

3.1. Analytical solution using one-term approximation

For the road profile defined by Eq. (5), the solution for Eq. (9) is sought in the form

$$\begin{aligned} y_u(\tau) &= Au_0 + Au_1 \cos(\omega\tau) + Bu_1 \sin(\omega\tau) \\ y_s(\tau) &= As_0 + As_1 \cos(\omega\tau) + Bs_1 \sin(\omega\tau) \end{aligned} \tag{10}$$

Note that the constant term is necessary for both y_u and y_s because the asymmetrical non-smooth nonlinearity introduced by the damper appears in the equations for both of them. Substituting this into Eq. (9) results in

$$\begin{aligned} Cu_0 + Cu_1 \cos(\omega\tau) + Cu_2 \sin(\omega\tau) + HOH &= 0 \\ Cs_0 + Cs_1 \cos(\omega\tau) + Cs_2 \sin(\omega\tau) + HOH &= 0 \end{aligned} \tag{11}$$

in which HOH, standing for higher order harmonics, are a result of the asymmetrical damping term $2\zeta_s^\pm y_s'$. While writing Eq. (11), the asymmetric damping term $2\zeta_s^\pm y_s'$, after substituting $y_s(\tau)$ from Eq. (10), has been replaced by its Fourier series expansion as

$$2\zeta_s^\pm y_s' = c_0 + \sum_{j=1}^{\infty} c_{2j-1} \sin(j\omega\tau) + \sum_{j=1}^{\infty} c_{2j} \cos(j\omega\tau). \tag{12}$$

The coefficients c_0 , c_{2j-1} and c_{2j} are given by

$$\begin{aligned} c_0 &= \frac{\omega}{2\pi} \int_0^{\frac{2\pi}{\omega}} 2\zeta_s^\pm \omega (-As_1 \sin(\omega\tau) + Bs_1 \cos(\omega\tau)) d\tau \\ c_{2j-1} &= \frac{\omega}{\pi} \int_0^{\frac{2\pi}{\omega}} 2\zeta_s^\pm \omega (-As_1 \sin(\omega\tau) + Bs_1 \cos(\omega\tau)) \sin(j\omega\tau) d\tau \\ c_{2j} &= \frac{\omega}{\pi} \int_0^{\frac{2\pi}{\omega}} 2\zeta_s^\pm \omega (-As_1 \sin(\omega\tau) + Bs_1 \cos(\omega\tau)) \cos(j\omega\tau) d\tau \end{aligned} \tag{13}$$

Note that the evaluation of the integrals above requires finding the time instants corresponding to transition from the compression phase ($y_s' < 0$), with the damping coefficient as ζ_s^- , to the expansion phase ($y_s' \geq 0$), with the damping coefficient as ζ_s^+ , and vice versa. The transition times are given accordingly by the two roots of the equation $y_s' = 0$, lying between 0 and $2\pi/\omega$. Under the one-term approximation for $y_s(\tau)$ as given by Eq. (10), the relevant equation is

$$-As_1 \sin(\omega\tau) + Bs_1 \cos(\omega\tau) = 0 \tag{14}$$

whose relevant roots are

$$\tau_c = \frac{1}{\omega} \arctan\left(\frac{Bs_1}{As_1}\right) \quad \text{and} \quad \tau_c = \frac{1}{\omega} \left[\arctan\left(\frac{Bs_1}{As_1}\right) \pm \pi \right] \tag{15}$$

The choice of the sign in the second root depends on the signs of As_1 and Bs_1 , so as to ensure that the roots lie in the right interval. For example, if both As_1 and Bs_1 are positive, the first root lies in the first quadrant and the second root, with a + sign, lies in the third quadrant. If both As_1 and Bs_1 are negative, the first root is in the third quadrant, while the second root, with a - sign, is in the first quadrant. Note that the above solution for τ_c implies that the damper spends the same amount of time π/ω during the expansion and the compression phases. This is not necessarily true. Hence, there is a need for an approximation involving more harmonics. However, the one-term approximation as above helps in getting a closed form analytical approximation for the periodic solution, and is still of interest in understanding the key qualitative features of the solution. Accordingly, the relevant Fourier coefficients, i.e. c_0 , c_1 and c_2 , are evaluated. Without loss of generality, it is assumed that the damper is in the compression phase at the beginning of the cycle. The coefficient c_0 can be evaluated as

$$c_0 = \frac{\omega^2}{\pi} \left[\int_0^{\tau_c} \zeta^- (-As_1 \sin(\omega\tau) + Bs_1 \cos(\omega\tau)) d\tau \tag{16}$$

$$+ \int_{\tau_c}^{\tau_c + \frac{\pi}{\omega}} \zeta^+ (-As_1 \sin(\omega\tau) + Bs_1 \cos(\omega\tau)) d\tau \tag{16}$$

$$\left. + \int_{\tau_c + \frac{\pi}{\omega}}^{\frac{2\pi}{\omega}} \zeta^- (-As_1 \sin(\omega\tau) + Bs_1 \cos(\omega\tau)) d\tau \right] \tag{16}$$

which reduces to

$$c_0 = \frac{2\omega}{\pi} (\zeta^- - \zeta^+) [As_1 \cos(\omega\tau_c) + Bs_1 \sin(\omega\tau_c)]. \tag{17}$$

In the above, the root for τ_c from Eq. (15) with $\omega\tau_c < \pi$ should be used. Similarly, the coefficients c_1 and c_2 can be evaluated as

$$c_1 = -\omega(\zeta^- + \zeta^+)As_1 \quad \text{and} \quad c_2 = \omega(\zeta^- + \zeta^+)Bs_1. \tag{18}$$

Note again that, under the one-term approximation, the coefficients c_1 and c_2 do not depend on the transition time τ_c . They also never involve As_0 , because of the use of the derivative and, therefore, the equations for evaluating the unknown coefficients As_1 , Bs_1 , Au_1 and Bu_1 for the approximation is essentially decoupled from the equations for determining the unknowns As_0 and Au_0 .

For Eq. (11) to be valid for arbitrary τ , it is required that the Cs_i 's as well as Cu_i 's vanish, yielding the following system of equations

$$kAu_0 - \frac{2\omega}{\pi} (\zeta_s^- - \zeta_s^+) [As_1 \cos(\omega\tau_c) + Bs_1 \sin(\omega\tau_c)] - As_0 = 0, \tag{19}$$

$$\left(k - \frac{\omega^2}{m}\right)Bu_1 - Bs_1 - 2\zeta_u \omega Au_1 + \omega(\zeta^- + \zeta^+)As_1 - \frac{\omega^2}{m}A = 0, \tag{20}$$

$$\left(k - \frac{\omega^2}{m}\right)Au_1 - As_1 + 2\zeta_u \omega Bu_1 - \omega(\zeta^- + \zeta^+)Bs_1 = 0, \tag{21}$$

$$-mkAu_0 - \frac{2\omega}{\pi}(\zeta_s^- - \zeta_s^+)(1+m)[As_1 \cos(\omega\tau_c) + Bs_1 \sin(\omega\tau_c)] + (1+m)As_0 = 0, \tag{22}$$

$$(1+m-\omega^2)Bs_1 - mkBu_1 - \omega(\zeta^- + \zeta^+)(1+m)As_1 + 2m\zeta_u \omega Au_1 = 0, \tag{23}$$

$$(1+m-\omega^2)As_1 - mkAu_1 + \omega(\zeta^- + \zeta^+)(1+m)Bs_1 - 2m\zeta_u \omega Bu_1 = 0. \tag{24}$$

Eqs. (20), (21), (23) and (24) can be solved for As_1 , Bs_1 , Au_1 and Bu_1 as

$$As_1 = \frac{-m\omega^3 A[k(\zeta^- + \zeta^+)(-m\omega^2 - \omega^2 + mk) + 4m\zeta_u^2 \omega^2(\zeta^- + \zeta^+) + 2\zeta_u \omega^2(1+m-\omega^2)]}{\omega^8 + a_6 \omega^6 + a_4 \omega^4 + a_2 \omega^2 + k^2 m^2} \tag{25}$$

$$Bs_1 = \frac{-m\omega^2 A[k(-\omega^4 + \omega^2(km + m + 1) - km) + 4m\zeta_u^2 \omega^2(\omega^2 - 1) + 2(1+m)\omega^4(\zeta^- + \zeta^+)]}{\omega^8 + a_6 \omega^6 + a_4 \omega^4 + a_2 \omega^2 + k^2 m^2} \tag{26}$$

$$Au_1 = \frac{-m\omega^3 A[km\omega^2(\zeta^- + \zeta^+) + 2\zeta_u(\omega^4 + \omega^2((1+m)(\zeta^- + \zeta^+)^2 - m - 2) + m + 1)]}{\omega^8 + a_6 \omega^6 + a_4 \omega^4 + a_2 \omega^2 + k^2 m^2} \tag{27}$$

and

$$Bu_1 = \frac{-\omega^2 A[\omega^6 + b_4 \omega^4 + b_2 \omega^2 - km(1+m)]}{\omega^8 + a_6 \omega^6 + a_4 \omega^4 + a_2 \omega^2 + k^2 m^2} \tag{28}$$

in which

$$a_6 = (1+m)^2(\zeta^- + \zeta^+)^2 - 2(1+m) + 4m^2\zeta_u(\zeta_u + \zeta^- + \zeta^+) - 2mk,$$

$$a_4 = (1+m)^2 + 4m^2\zeta_u^2((\zeta^- + \zeta^+)^2 - 2) + km(2(1+m)(1 - (\zeta^- + \zeta^+)^2) + km + 2),$$

$$a_2 = m^2(k((\zeta^- + \zeta^+)^2 k - 2k - 2) + 4\zeta_u^2) - 2km,$$

$$b_4 = m^2(\zeta^- + \zeta^+)(2\zeta_u + \zeta^- + \zeta^+) + m(2(\zeta^- + \zeta^+)^2 - k - 2) + (\zeta^- + \zeta^+)^2 - 2,$$

and

$$b_2 = m^2(k - k(\zeta^- + \zeta^+)^2 + 1) + m(2k - k(\zeta^- + \zeta^+)^2 + 2) + 1.$$

From the above solutions for Au_1 , Bu_1 , As_1 and Bs_1 , it is difficult to make any judgement about the right sign for τ_c given in Eq. (15) for different ranges of ω . However, one should just remember to use the lower value of τ_c to get the mean shift As_0 and Au_0 from Eqs. (19) and (22). From Eqs. (19) and (22), one can easily find that

$$Au_0 = 0.$$

The solution for the mean term As_0 , introduced due to the asymmetrical damper, after duly accounting for the right τ_c is

$$As_0 = \frac{2m\omega^3 \sqrt{4\zeta_u^2 \omega^2 + k^2}(\zeta^- - \zeta^+)A}{\pi \sqrt{\omega^8 + a_6 \omega^6 + a_4 \omega^4 + a_2 \omega^2 + k^2 m^2}} \tag{29}$$

in which the constants a_6 , a_4 and a_2 are as reported above. Note that $As_0 < 0$ when $\zeta^+ > \zeta^-$, while $As_0 > 0$ when $\zeta^- < \zeta^+$. In general, for vehicle dynamics applications, $\zeta^+ > \zeta^-$, which leads to $A_0 < 0$, referred to as jacking down of the suspension. The final one-term approximation for the absolute displacement of the suspended mass z_s under harmonic road profile is given by

$$z_s(\tau) = z_r + y_u + y_s \\ = As_0 + (As_1 + Au_1) \cos(\omega\tau) + (A + Bs_1 + Bu_1) \sin(\omega\tau) \tag{30}$$

with the constants As_0 , As_1 , Au_1 , Bs_1 and Bu_1 given by Eqs. (25–29).

With this analytical form for the approximation and an analytical solution for the switching time τ_c , the smaller of the two roots in Eq. (15), it is possible to calculate all relevant quantities of interest, such as the energy dissipated by the damper during the compression and the expansion phases. Note that the one-term approximation presented in this subsection implies that the system spends the same time during the compression and the expansion phases, and the energies as calculated from the above analytical approximation are almost the same. To account for the different times spent by the system during the two phases, higher order harmonics are needed, which is presented next.

3.2. Approximation with higher harmonics

With the higher harmonics, the solution for Eq. (9) takes the form

$$y_s(\tau) = As_0 + \sum_{n=1}^N [As_n \cos(n\omega\tau) + Bs_n \sin(n\omega\tau)] \tag{31}$$

and

$$y_u(\tau) = Au_0 + \sum_{n=1}^N [Au_n \cos(n\omega\tau) + Bu_n \sin(n\omega\tau)], \tag{32}$$

in which N represents the number of harmonics retained in the approximation. With this approximation, the transition times from the compression to the expansion phase, and vice versa, are given by the first two non-zero roots (τ_{c1} , τ_{c2} , with $\tau_{c1} < \tau_{c2}$), of the equation

$$\sum_{n=1}^N [-nAs_n \sin(n\omega\tau) + nBs_n \cos(n\omega\tau)] = 0. \tag{33}$$

Again without loss of generality, it is assumed that the cycle starts with the compression phase. Accordingly, the various terms of interest in the Fourier series of the asymmetrical damping term

$$2 \zeta_s^\pm y_s' = c_0 + \sum_{j=1}^N c_{2j-1} \sin(j\omega\tau) + \sum_{j=1}^N c_{2j} \cos(j\omega\tau) \tag{34}$$

can be calculated as

$$\begin{aligned} c_0 = & \frac{\omega^2}{\pi} \left[\int_0^{\tau_{c1}} \zeta_s^- \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) d\tau \right. \\ & + \int_{\tau_{c1}}^{\tau_{c2}} \zeta_s^+ \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) d\tau \\ & \left. + \int_{\tau_{c2}}^{\frac{2\pi}{\omega}} \zeta_s^- \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) d\tau \right], \end{aligned} \tag{35}$$

$$\begin{aligned} c_{2j-1} = & \frac{2\omega^2}{\pi} \left[\int_0^{\tau_{c1}} \zeta_s^- \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) \sin(j\omega\tau) d\tau \right. \\ & + \int_{\tau_{c1}}^{\tau_{c2}} \zeta_s^+ \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) \sin(j\omega\tau) d\tau \\ & \left. + \int_{\tau_{c2}}^{\frac{2\pi}{\omega}} \zeta_s^- \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) \sin(j\omega\tau) d\tau \right] \end{aligned} \tag{36}$$

and

$$\begin{aligned} c_{2j} = & \frac{2\omega^2}{\pi} \left[\int_0^{\tau_{c1}} \zeta_s^- \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) \cos(j\omega\tau) d\tau \right. \\ & + \int_{\tau_{c1}}^{\tau_{c2}} \zeta_s^+ \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) \cos(j\omega\tau) d\tau \\ & \left. + \int_{\tau_{c2}}^{\frac{2\pi}{\omega}} \zeta_s^- \sum_{n=1}^N n(-As_n \sin(n\omega\tau) + Bs_n \cos(n\omega\tau)) \cos(j\omega\tau) d\tau \right] \end{aligned} \tag{37}$$

using $j = 1 : N$. Substituting $y_s(\tau)$ from Eq. (31), $y_u(\tau)$ from Eq. (32) and the Fourier series expansion for the asymmetrical damper from Eq. (34) in Eq. (9), it is possible to write

$$\begin{aligned}
 Cu_0 + \sum_{n=1}^N [Cu_{2n-1} \cos(n\omega\tau) + Cu_{2n} \sin(n\omega\tau)] + HOH &= 0 \\
 Cs_0 + \sum_{n=1}^N [Cs_{2n-1} \cos(n\omega\tau) + Cs_{2n} \sin(n\omega\tau)] + HOH &= 0
 \end{aligned}
 \tag{38}$$

in which *HOH* stands for the harmonics of order higher than N of the asymmetrical damping term $2\zeta_s^\pm y_s'$. Again, validity of Eq. (38) for arbitrary τ is achieved by requiring the various Cu_i 's and Cs_i 's to vanish. This gives $4N + 2$ equations, which in conjunction with two equations for τ_{c1} and τ_{c2} , obtained by substituting $\tau_c = \tau_{c1}$ and $\tau_c = \tau_{c2}$ in Eq. (33), gives the complete set of $4N + 4$ equations in the $4N + 4$ unknowns, viz. τ_{c1} , τ_{c2} , As_0 , Au_0 , As_n , $n = 1, \dots, N$, Bs_n , $n = 1, \dots, N$, Au_n , $n = 1, \dots, N$, and Bu_n , $n = 1, \dots, N$. It is emphasized here again that under the approximation with the higher harmonics, all the Fourier coefficients depend on the transition times, making the coupling stronger, and also causing the entire system of equations to be nonlinear, as opposed to the one-term approximation in which only the coefficient c_0 depends on τ_c , while the coefficients c_1 and c_2 were independent of τ_c . However, the equations are linear in the unknown coefficients of the approximation As_0 , Au_0 , As_n , Au_n , Bs_n and Bu_n , and the only source of nonlinearity comes from the unknown transition times τ_{c1} and τ_{c2} . The unknown coefficients As_0 , Au_0 , As_n , Au_n , Bs_n and Bu_n can be solved in closed form in terms of the forcing parameters A and ω , the system parameters m , k , ζ_u , ζ_s^+ and ζ_s^- , and the transition times τ_{c1} and τ_{c2} . Substitution of these in Eq. (33) results in a set of coupled equations for the transition times τ_{c1} and τ_{c2} , which have to be solved numerically for each combination of the system and forcing parameters. However, since the nonlinearity has the scaling property (i.e., $f(ax) = a f(x)$), these equations become independent of the forcing amplitude A , and depend only on the forcing frequency ω and the damping coefficients ζ_u , ζ_s^+ and ζ_s^- , along with the mass and stiffness ratios, m and k .

Once the transitions times τ_{c1} and τ_{c2} are evaluated numerically, the approximation for the periodic solution of the system is known in closed form, and again all the relevant quantities of interest can be evaluated. The inclusion of the higher harmonics also results in the possibility of evaluating the different times which the system spends in the compression and the expansion phases. The time spent in the expansion phase is $\tau_{c2} - \tau_{c1}$, while the time spent during the compression phase is $2\pi/\omega + \tau_{c1} - \tau_{c2}$.

4. Convergence of approximate analytical solutions

For illustration of the differences between the solutions from the one-term approximation and the approximation with higher harmonics, the parameters are set to $k = \frac{100}{11}$, $m = \frac{15}{2}$, $\zeta_u = \frac{\sqrt{66}}{3300}$, $\zeta_s^+ = \frac{\sqrt{66}}{22}$, $\zeta_s^- = \frac{1}{\sqrt{66}}$, $\omega = \frac{5\sqrt{66}}{11}$ and $A = 1$. All parameters are representative of a medium passenger vehicle (e.g. a sedan carrying four passengers) [3,4,15,24]. The choice of A is not really important due to the scalability property mentioned earlier. The solutions for the transition times are independent of A , and the solutions for the unknown coefficients A_0 , A_n and B_n can simply be obtained by multiplying the solution for $A = 1$ by the desired A value. For this set of parameters, the solution for the one-term approximation is

$$\begin{aligned}
 y_s(\tau) &= -0.6753929944 - 1.160029991 \sin(\omega\tau) + 0.1273125741 \cos(\omega\tau) \\
 y_u(\tau) &= 0.05798611519 \sin(\omega\tau) - 0.2726469840 \cos(\omega\tau)
 \end{aligned}
 \tag{39}$$

with the transition time from compression phase to expansion phase as 0.4549754579, while the transition time from expansion to compression phase is 1.305722770. The time-period is 1.701494625, and the system spends 0.8507473122 in both the expansion and the compression phases. The total displacement of the suspended mass is

$$z_s(\tau) = -0.6753929944 - 0.1020438758 \sin(\omega\tau) - 0.1453344099 \cos(\omega\tau) .
 \tag{40}$$

For the two-term approximation, the solution is

$$\begin{aligned}
 y_s(\tau) &= -0.6781289249 - 1.155693927 \sin(\omega\tau) + 0.1228181574 \cos(\omega\tau) \\
 &\quad - 0.0842603988 \sin(2\omega\tau) - 0.04093687024 \cos(2\omega\tau)
 \end{aligned}
 \tag{41}$$

and

$$\begin{aligned}
 y_u(\tau) &= 0.04866710965 \sin(\omega\tau) - 0.2630427168 \cos(\omega\tau) \\
 &\quad + 0.08713668254 \sin(2\omega\tau) + 0.04240858075 \cos(2\omega\tau)
 \end{aligned}
 \tag{42}$$

with the transition times $\tau_{c1} = 0.4130503940$ and $\tau_{c2} = 1.332931751$. The system spends 0.9198813570 time units in the expansion phase, which is slightly higher than 0.7816132680 which it spends in the compression phase. The total displacement of the suspended mass is

$$\begin{aligned}
 z_s(\tau) &= -0.6781289249 - 0.1070268174 \sin(\omega\tau) - 0.1402245594 \cos(\omega\tau) \\
 &\quad + 0.00287628372 \sin(2\omega\tau) + 0.00147171051 \cos(2\omega\tau)
 \end{aligned}
 \tag{43}$$

Table 1
Switching times with higher harmonics.

N	τ_{c1}	τ_{c2}
1	0.4549754579	1.305722770
2	0.4130503940	1.332931751
3	0.4003939010	1.334942409
4	0.4016793221	1.337299495
5	0.4000764645	1.336891773
6	0.4004410651	1.337329595
7	0.4000541553	1.337249220

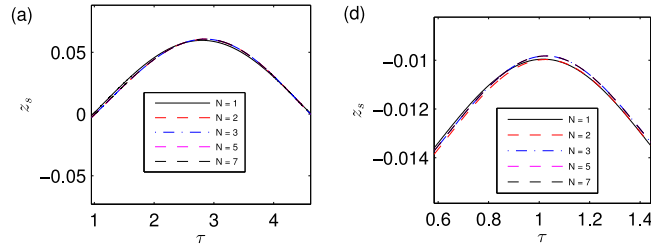


Fig. 2. Displacement of the suspended mass for various numbers of harmonics at $\omega = 5$ (a) and $\omega = 20$ (b).

Note that the values for $As_0, Au_0, As_1, Au_1, Bs_1$ and Bu_1 as obtained from the one-term approximation as well as the two-term approximation are very close. Also, the magnitude of the coefficients of the second harmonics are at least an order smaller than the coefficients of the terms of the first harmonics, indicating convergence. For this set of parameters, the time spent in the compression and the expansion phases are different, but the difference is not very large and hence, a one-term approximation yielding closed form solutions is a good approximation. For values closer to the resonance, i.e., $\omega = 1$, an approximation with higher harmonics is required, because in this regime, the time spent by the system during compression and expansion phases is significantly different. To complete this analysis of the convergence of the approximation, the solution with three harmonics is presented as well, which is

$$\begin{aligned}
 y_s(\tau) = & -0.6715757140 - 1.155735330 \sin(\omega\tau) + 0.122548469 \cos(\omega\tau) \\
 & - 0.09005465407 \sin(2\omega\tau) - 0.04214340498 \cos(2\omega\tau) \\
 & - 0.01250716472 \sin(3\omega\tau) + 0.00891232319 \cos(3\omega\tau)
 \end{aligned} \tag{44}$$

and

$$\begin{aligned}
 y_u(\tau) = & 0.0487541786 \sin(\omega\tau) - 0.2624645675 \cos(\omega\tau) \\
 & + 0.09312987609 \sin(2\omega\tau) + 0.04366085332 \cos(2\omega\tau) \\
 & + 0.01181099136 \sin(3\omega\tau) - 0.008408776026 \cos(3\omega\tau)
 \end{aligned} \tag{45}$$

The total displacement of the suspended mass under this approximation is

$$\begin{aligned}
 z_s(\tau) = & -0.6715757140 - 0.1069811513 \sin(\omega\tau) - 0.1399160976 \cos(\omega\tau) \\
 & + 0.00307522202 \sin(2\omega\tau) + 0.00151744834 \cos(2\omega\tau) \\
 & - 0.00069617336 \sin(3\omega\tau) + 0.00050354716 \cos(3\omega\tau)
 \end{aligned} \tag{46}$$

The switching times for the three-term approximation are 0.4003939010 and 1.334942409, which differ from the ones from the two-term approximation in the second decimal place. To further investigate the convergence aspect, the variation in the transition times with up to $N = 7$ is summarised in Table 1.

From these results, it is possible to safely conclude that an approximation with 4 harmonics is capturing the solution very well for this set of parameters. For values of ω close to 1, one might require more harmonics to capture the transition times accurately. However, as far as the total displacement of the suspended mass is concerned, even the one-term approximation is very good, as shown in Fig. 2.

5. Influence of the asymmetry ratio

Using the parameter values introduced in the previous section, the influence of β is studied for different values of ω . Fig. 3 shows the maximum displacement, maximum velocity and maximum acceleration of the suspended mass for varying ω . Three possibilities for values of β are used, one representing symmetrical damping ($\beta = 1$), one representing asymmetrical damping with $c^+ > c^-$ ($\beta = 3$), and one representing asymmetrical damping with $c^+ < c^-$ ($\beta = 1/3$). Although asym-

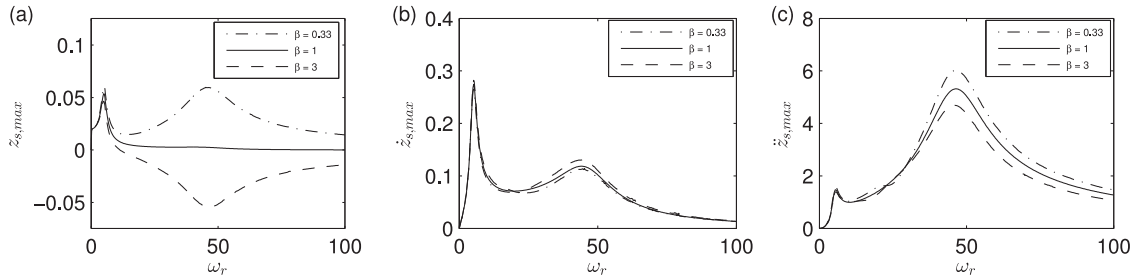


Fig. 3. Maximum displacement (a), maximum velocity (b) and maximum acceleration of the suspended mass for varying excitation frequency with three values of β .

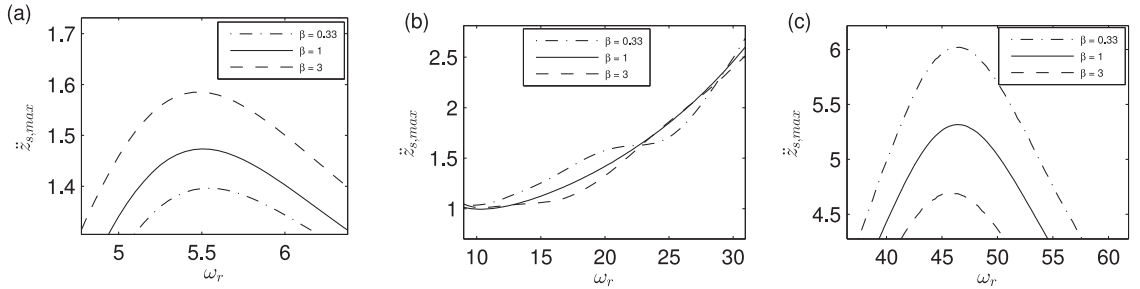


Fig. 4. Maximum acceleration of the suspended mass close to the first resonance peak (a), at intermediate frequencies (b) and close to the second resonance peak (c).

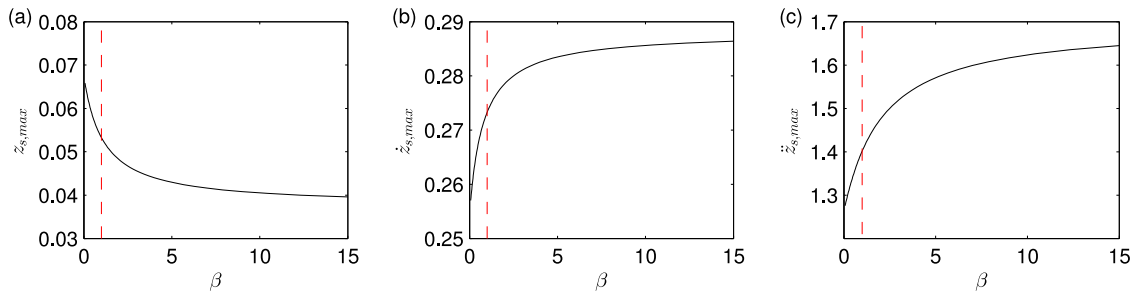


Fig. 5. Maximum displacement (a), maximum velocity (b) and maximum acceleration of the suspended mass for varying asymmetry ratio with $\omega_r = 5.41$ rad/s.

metrical dampers in vehicle systems usually have β close to 3 [16], it is shown in Fig. 3 that $\beta < 1$ results in smaller accelerations for $0 < \omega_r < 20$ rad/s.

Comparing the accelerations at the two resonance peaks (Fig. 4), the use of $\beta < 1$ presents a reduction of 5.29 % at the first peak, while resulting in an increase of 13.21 % at the second peak when compared to the symmetrical case. The use of $\beta > 1$ presents an increase of 7.60 % at the first peak, and a reduction of 13.25 % at the second peak. For both cases of β , the difference from the symmetrical case diminishes as ω_r increases past 100 rad/s.

Fixing the excitation frequency at the first resonance level, it is possible to see that acceleration values are lower for $\beta < 1$ (Fig. 5). This behaviour reverts at the second resonance level, at which the acceleration values are lower for $\beta > 1$ (Fig. 6). Note that this behaviour is dependent on the direction of the acceleration, which is important for comfort perception [25,26]. If the absolute value of acceleration is considered, the performance with $\beta < 1$ or $\beta > 1$ presents advantage over the symmetrical case in the same frequency ranges.

The ratio of energy dissipated during the compression phase and the energy dissipated during the expansion phase is given by

$$E_R = \frac{\int_0^{\tau_{c1}} \zeta_s^- \dot{x}^2 dt + \int_{\tau_{c2}}^{2\pi/\omega} \zeta_s^- \dot{x}^2 dt}{\int_{\tau_{c1}}^{\tau_{c2}} \zeta_s^+ \dot{x}^2 dt} \tag{47}$$

while the ratio of time spent in the compression phase and time spent in the expansion phase is given by

$$\tau_R = \frac{\tau_{c1} + 2\pi/\omega - \tau_{c2}}{\tau_{c2} - \tau_{c1}} \tag{48}$$

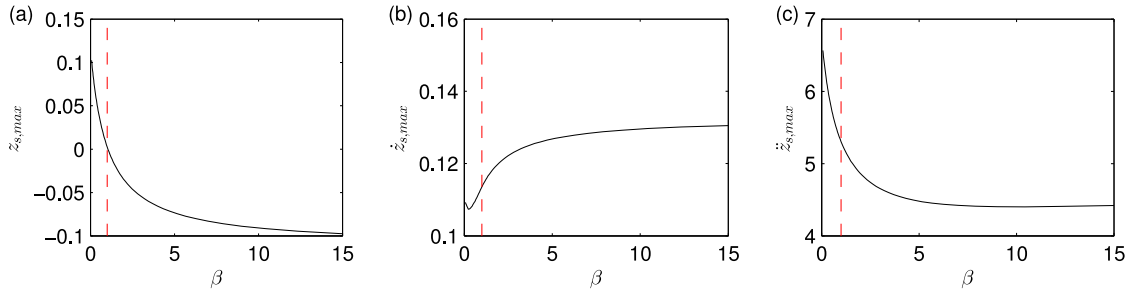


Fig. 6. Maximum displacement (a), maximum velocity (b) and maximum acceleration of the suspended mass for varying asymmetry ratio with $\omega_r = 44.72$ rad/s.

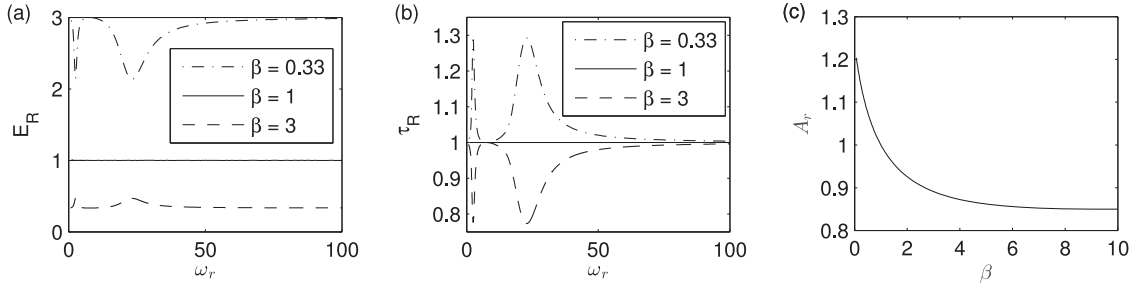


Fig. 7. Energy (a) and time (b) ratios for varying ω_r ; and ASD ratio for varying β (c).

Fig. 7 shows these ratios as function of the excitation frequency. Close to the two resonance peaks, the time spent in the expansion phase is larger with $\beta > 1$, while the time spent in the expansion phase is larger with $\beta < 1$. However, due to the asymmetrical damping, the magnitude of velocities is higher during the shorter phases, resulting in the opposite behaviour of the energy ratio. Also shown is the asymmetrical to symmetrical ratio of acceleration spectral density (A_r) of acceleration responses for varying β . Where $A_r < 1$, the asymmetrical system has a net advantage over the whole frequency range.

6. Conclusions

In the analysis of the suspension system using symmetrical and asymmetrical damping with harmonic excitation, it was observed that the asymmetry ratio directly influences the maximum displacement, maximum speed and maximum acceleration of the sprung mass. Regarding the maximum acceleration, the asymmetrical damping proves more advantageous at higher frequencies with $\beta > 1$. In the region of lower frequencies, $0 < \omega_r < 20$ rad/s, the asymmetrical system with $\beta < 1$ proves more advantageous. Although the average position of the sprung mass varies, the displacement amplitude practically remains. The jacking down effect is caused by the asymmetrical damping characteristics, but the frequency of the harmonic excitation influences the average position of the suspended system.

The difference in acceleration amplitude between the symmetrical and asymmetrical cases is explained by the different dissipation of energy in the system in the expansion and compression phases, and also with the different time spent in each phase, with more time in the phase with lower damping.

Although current systems usually adopt $\beta > 1$, it is shown in this work that, for lower frequencies, $\beta < 1$ results in better comfort. This observation can be used for devising a control system to tune the value of β depending on the road and driving conditions.

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